



Probabilistic Damage Tolerance Analysis Using Adaptive Multiple Importance Sampling

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Probabilistic Risk Assessment

- Probabilistic Risk Assessment is an important tool for ensuring structural integrity of aircraft components.
- Based on the principles of probabilistic damage tolerance analysis.
- The Single Flight Probability-of-Failure is difficult to compute accurately and efficiently due to several challenges:
 - Very small probabilities, e.g., 1E-7 or smaller
 - Standard Monte Carlo sampling is impractical
 - Inspection and repair process results in multi-modal crack size distributions
 - FORM/SORM methods are impractical
 - > Inspection optimization requires multiple analyses
 - Efficient reanalyses are required







10⁻⁵

10⁻⁶

With Inspections
Without Inspections

Probabilistic Risk Assessment





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Probability of Failure Calculation



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 $POF(t) = P[\sigma_{MAX} > \sigma_{RS}(t)]$



Probability Equations



The probability-of-failure is the probability that maximum value of the applied stress (during the next flight) will exceed the residual strength σ_{RS} of the aircraft component.

$$POF_{Lincoln}(t) = P[\sigma_{Max} > \sigma_{RS}(t)] = \int [1 - F_{EVD}(\sigma_{RS}(\mathbf{x}, t_n))] f_{X}(\mathbf{x}) dx$$

$$\sigma_{Max} > \sigma_{RS}$$
 Other random variables

Survival to time t

$$POF_{Freudenthal}(t_n) = \frac{\int \left[\prod_{i=1}^{n-1} F_{EVD}(\sigma_{RS}(\boldsymbol{x}, t_i))\right] [1 - F_{EVD}(\sigma_{RS}(\boldsymbol{x}, t_n)] f_X(\boldsymbol{x}) d\boldsymbol{x}}{\int \prod_{i=1}^{n} F_{EVD}(\sigma_{RS}(\boldsymbol{x}, t_i)) f_X(\boldsymbol{x}) d\boldsymbol{x}}$$

 F_{EVD} - CDF of the maximum stress per flight (extreme value distribution) σ_{RS} - residual strength



SMART |DT





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Importance Sampling











- > Goal is to locate important regions in standard normal space.
- > Generate samples near and around the important regions for all evaluation times.
- > Performs exploration to find the location of important regions.
- The adaptation phase will focus on determining the scale and shape of important regions.







- Determine evaluation times at which to focus samples. Note, near-by times also obtain improved results.
- Use Coefficient of Variation (COV) which is a normalized error estimate.
- Ensures COV across all evaluation times is below a user-defined threshold.



Adaptive Multiple Importance Sampling Approach



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- Approximate the averaged or combined important region using a mixture density composed of multivariate normal sampling densities optimized for individual evaluation times.
- > Key advantage is that samples can be used for more than one important region where regions overlap.

$$E[H(x,t)] = \frac{1}{N_{mix} N_{samp}} \sum_{j=1}^{N_{mix}} \sum_{i=1}^{N_{samp}} H(\mathbf{x}_{ij},t) \frac{f(\mathbf{x}_{ij})}{(1/N_{mix}) \sum_{k=1}^{N_{mix}} q(\mathbf{x}_{ij},\boldsymbol{\theta}_k)}$$

N.Crosby, "Efficient Adaptive Importance Sampling Estimation of Time Dependent Probability of Failure with Inspections for Damage Tolerant Aircraft Structures," PhD dissertation, University of Texas at San Antonio, 2021







- Risk assessment handbook example using a closed-form crack growth equation.
- > General aviation example with inspections.



Risk Assessment Handbook Problem





Parameter	Value
Width	Deterministic 10 in
Radius	Deterministic 0.125 in
Initial Crack Size	<i>LN</i> (0.0032, 0.0047) in
Fracture Toughness	N(34.8, 3.90) ksi √in
Maximum Stress per Flight	<i>W</i> (5.0,10.0, 5.0) ksi

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Tuegel et al., Aircraft structural reliability and risk analysis handbook volume 1: Basic analysis methods., Technical report, Air Force Research Lab, Wright-Patterson AFB, OH, Aerospace Systems Dir, 2013







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Independent verification

		AI Li	MIS Lincoln (880 s ao 2012, Lincoln	amples)	AMIS Freuden Liao 2012, Fre	ithal (3040 sam eudenthal	ples)	15 evaluation times
SFPOF	10^{-4}							COV threshold 0.1
	10^{-5}							Lincoln Formulation
	10^{-6} 10^{-7}							(assumes survival = 1 from flight 0 to flight t)
	10 ⁻⁸							80 samples per iteration
	10 ⁻⁹							11 iterations
	10 ⁻¹⁰							> 880 samples
	10 ⁻¹¹							Freudenthal Formulation
	1.0 0.8							(does not assume survival = 1 from flight 0 to flight t)
	0.6 0.4							160 samples per iteration
	0.2							19 iterations
	0.0	0	2000	4000	6000	8000	10000	> 3040 samples
				Flight ho	urs			

Liao M., Comparison of different single flight probability of failure (SFPOF) calculations for aircraft structural risk analysis. In Aircraft Airworthiness and Sustainment (AA&S) Conference, 2012





- Variations calculated for 100 PDTA AMIS runs.
- For both Lincoln and Freudenthal POF Formulations.
 - PDTA AMIS estimates are within the expected error bands, showing the sampling variance gives a good indication of estimator error.
 - > PDTA AMIS median error is close to 0, showing the estimates are consistent.

POF Inspections





- Inspections are not deterministic there is some probability of missing cracks
- In PDTA, this is modeled by reducing the probability of failure proportional to undetected cracks
- > PND is the probability of not detecting a crack in any inspection(s) before t



Change in Combined Important Region Due to Inspection



Combined Important Region without inspection

Combined Important Region with one Inspection



- > Post-inspection, a new important region emerges around ($a_i = 0.007$, $k_c = 12.5$).
- Stored crack growth analyses reevaluated with the modified response function including an inspection provide a good general idea of the new important region location



cost

х

3

2

1

After each PDTA AMIS run:

10000

150 1.0 t

0.8

0.2

≥ 0.6 0 0.4

≥ 0.6 0.4

0.2 0.0

> Update conditional POF, $H(\cdot)$, to include new inspection time in PND function

10-

HS 10-8

150 1.0 t 0.8 0.6

> 0.2 0.0

POF

10000

Inspection 3

Flight hours

- > Recalculate $H(\cdot)$ for all samples over all times with existing crack growth evaluations
- Re-run PDTA AMIS adaptation

PDTA AMIS only has to add crack growth evaluations to adapt for the new inspection

No. additional inspections

- SMC must rerun all of the crack growth evaluations
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slows down with

additional inspections

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General Aviation Example Problem



Parameter	Values
Width	Deterministic 5 in
Thickness	Deterministic 0.125 in
Log Paris Constant	N(-9.0, 0.08)
Paris Exponent	Deterministic 3.8
Initial Crack Size	W(0.45, 4.17
	$\times 10^{-5}$) in
Fracture Toughness	$N(35.0, 3.5)$ ksi \sqrt{in}
Maximum Stress per Flight	<i>EVD</i> (13.4, 1.3, 0.07) ksi
Probability of Detection	<i>LN</i> (0.05, 0.065) in
Repair Quality (Crack Size)	Perfect



POF Results After Adding 8 Inspections



Flight hours





SFPOF



POF Results with Opt. Inspections

Risk-Threshold – Find the minimal set of inspections that maintains risk below a given threshold using a single inspection type

Keep risk below 1E-7



Juan Ocampo, Nathan Crosby, Harry Millwater, Chris Hurst, Beth Gamble, and Marv Nuss, "Fleet Management Considering Inspection Schedule Optimization", Aircraft Airworthiness & Sustainment Conf., Ponte Verde, FL, August 2022



POF Results with Opt. Inspections



Minimal-Cost – Find the minimum cost set of inspections that maintains risk below a given threshold using multiple inspection types

SMARTIDT	Information Analysis	Material Geometry	Loading Q	ons Run Resu	llts	Re
spection Schedule Type	Inspection Presets					G
Minimal Cost	Name	Туре	Inspection Prob.	Detection Prob.	Repaired Crack	
	AutomatedBoltHole		0.8	μ0.01799 σ0.01087 LN	Same as Original	
	EddyCurrentSlidingProbe EddyCurrentSlidingProbe	ddy Current Sliding Probe	0.8	μ0.0788 σ0.03022 LN	Same as Original	
		Edit Inspection Pro	eset			
		Name				
	Delete	AutomatedBoltHole				
	Risk Level	Deterministic Dist Material Custom	ributions Inspection Type	Geometry	Equipment	Summary
	Selected Inspection Presets	Aluminum				
	AutomatedBoltHole	Inconel 718 & Haynes 188				
	EddyCurrentSlidingProbe	Titanium				
		Probability of Detection	on Probat	ility of Inspection	Repaired Crack Size	Minimum I 1000.0
		LogNormal			Custom	Cost
		MEAN STDE	EV VALUE		Perfect	2.0
		0.01799 0.01	0.8		DISTRIBUTION	~
					SHAPE SCALE	









- The AMIS algorithm estimates POF for risk assessment using 6 orders of magnitude fewer samples compared to standard Monte Carlo sampling for probabilities of 10⁻⁷ with COV of 0.1.
- This efficiency allows for the incorporation of additional random variables into the problem and the use of more realistic fracture mechanics solutions.
- The SMART methodology in combination with AMIS is not limited to aircraft fleet COS and can also be applied to digital twin modeling, virtual testing, and other novel applications



Future Developments



- > Optimized inspection schedule
 - Determine the inspection times and inspection methods to keep the risk below a user-defined threshold with minimum cost.
- Bayesian Module Implementation for Sensor Integration
 - > Update crack size simulations with sensor (POD) data.
- Probabilistic damage tolerance analysis of more realistic structures
 - Continuing damage, multisite damage, residual stresses, out-of-plane crack growth, etc.
- > Approaches
 - NASGRO interface
 - Surrogate models
 - > Machine learning approaches, e.g., Bingo software, BAMF, etc.



Smart|DT Software



- Probabilistic risk assessment development has been funded by the US Federal Aviation Administration to develop the Smart|DT software.
- > Available to the general public.
- Training presented annually and available online:
 - Aircraft Airworthiness Conference
 - https://smartdtsoftware.wixsite.com/smart





Interoperability



Scriptable

SATE KIRES	
	87 080 6020 0.580 0.00



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Adaptive Multiple Importance Sampling Approach







Adaptive Multiple Importance Sampling Approach



- Approximate the averaged or combined important region using a mixture density composed of multivariate normal sampling densities optimized for individual evaluation times.
- > Key advantage is that samples can be used for more than one important region where regions overlap.





Academic Example





Random variables:

 \succ

 \triangleright

 \succ

- Initial crack size
- Fracture toughness
- Max stress per flight

- n 2 analysis times: t=0 and t=20000
- n No inspections

Initialization 1/8





- $\succ N_{mix} = 1$
- > Initial mixture set at the mean with covariance matrix $c_{\sigma}^2 I$.
- > 20 samples generated.

- Initialize empty mixture density, starting μ^* at origin, k^* points to last time in list of evaluation times.
- Add $(\mu^*, c_{\sigma}^2 I)$ to $\{\theta_{mix}\}$, set $\epsilon^* = \epsilon_{KL}/10$ to check that μ^* is not changing before moving to next evaluation time.
- Generate samples from $N(\mu^*, c_{\sigma}^2 I)$ and evaluate crack growth and response functions.



Initialization 2/8





- ⁿ Using the 20 samples, calculate μ^* using the cross-entropy method.
- Evaluate D_{min} from μ^* to all component densities in the mixture.



Initialization 3/8



Focus time t=20000



- > 20 new samples drawn from density #2.
- ➤ 40 samples total.

- n Add $(\mu^*, c_{\sigma}^2 I)$ to $\{\theta_{mix}\}$, set $\epsilon^* = \epsilon_{KL}/10$ to check that μ^* is not changing before moving to next evaluation time.
- ⁿ Generate samples from $N(\mu^*, c_{\sigma}^2 I)$ and evaluate crack growth and response functions.



Initialization 4/8



Focus time t=20000 5.0 2.5 Density rejected $D_{min} < \epsilon^*$ 0.0 <u>ک</u>" −2.5 -5.0-7.5 -10.0 --5.0 -2.5 0.0 2.5 7.5 10.0 5.0 a_0

- $\succ N_{mix} = 2$ prior densities.
- > All 40 samples used to compute new location μ^* .

$$\epsilon^{\star} = 0.1$$

 $\mu^{\star} = (1.5, -0.2)$
 $D_{min} = 0.08$

- $\succ D_{min} < \epsilon^*$
- Evaluation time satisfied. Moving to next time value.
- ➢ No samples generated since the new point (red) is close to another density $(D_{min} = 0.08)$.

- n Calculate μ^* using standard weights.
- Evaluate D_{min} from μ^* to all component densities in the mixture.



Initialization 5/8





- ≻ $N_{mix} = 2$ prior densities.
- > All 40 samples used to compute new location μ^* .

$$\epsilon^{\star} = 1.0$$

 $\mu^{\star} = (6.7, -1.2)$
 $D_{min} = 1.6$

- New density added centered at red cross.
- $\succ N_{mix} = 3$

- n The value *H* changes for the new time point.
- n Calculate μ^* using standard weights.
- Evaluate D_{min} from μ^* to all component densities in the mixture.
- n $D_{min} > \epsilon^*$ so focus on the next evaluation time and locate its important region.



Initialization 6/8





- > $N_{mix} = 3$ prior densities.
- Generate 20 new samples from density #3.
- > Now 60 samples total.

- n Add $(\mu^*, c_{\sigma}^2 I)$ to $\{\theta_{mix}\}$, set $\epsilon^* = \epsilon_{KL}/10$ to check that μ^* is not changing before moving to next evaluation time.
- n Generate samples from $N(\mu^*, c_{\sigma}^2 I)$ and evaluate crack growth and response functions.



Initialization 7/8





- n Calculate μ^* using standard weights.
- Evaluate D_{min} from μ^* to all component densities in the mixture.
- n $D_{min} < \epsilon^*$ so focus on the next evaluation time and locate its important region

N_{mix} = 3 prior densities.
 60 samples used to compute new location μ*.

$$\epsilon^{\star} = 0.1$$

 $\mu^{\star} = (6.5, -1.6)$
 $D_{min} = 0.01$

- > $D_{min} < \epsilon^*$. Evaluation time satisfied. Moving to next time value.
- > No samples generated since the new point (red) is close to another density $(D_{min} = 0.01)$



Initialization 8/8





- > Initialization complete.
- > 3 densities sufficient for initialization for 2 time points.
- > Total of 60 samples.
- Crack growth and POF values saved for every sample.

- n $k^* = 0$, so initialization routine is finished.
- n Return mixture density, realizations, crack growth evaluations and response function evaluations.



Adaptation Iteration 1/6





These equations are used to compute the POF and COV.



- n Update balance heuristic importance weights
- n Calculate estimates, estimator variances, and COVs
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached
- n Select time with the highest COV



Adaptation Iteration 1/6





≻60 samples used to compute a new density.

$$\mu^{\star} = \frac{\sum_{j} \sum_{i} h_{ijk} w_{std}(\boldsymbol{x}_{ij}) \boldsymbol{x}_{ij}}{\sum_{j} \sum_{i} h_{ijk} w_{std}(\boldsymbol{x}_{ij})}$$
$$\Sigma^{\star} = I$$

- New density added, $N_{mix} = 4$.
- > 20 new samples generated (80 total).

- n Calculate standard weights
- ⁿ Effective sample size is 1, insufficient to update covariance matrix. $\Sigma^* = I$.
- New density is $N(\mu^*, I)$, generate samples and evaluate crack growth and response function



Adaptation Iteration 2/6





- n Update balance heuristic importance weights
- n Calculate estimates, estimator variances, and COVs
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached
- n Select time with the highest COV

- POF and COV's computed from 80 samples.
- > Weights updated using all 80 samples.
- > All pofs and covs updated.
- COV reduced at t=0, moving to t=20,000.

$$w_{bh}(x_{ij}) = \frac{q(x_{ij}, \theta_j)}{\sum_{l=1}^{N_m} (1 / N_m) q(x_{ij}, \theta_l)}$$

$$P_f(t_k) = \frac{1}{N} \sum_j \sum_i h_{ijk} w_{bh}(x_{ij})$$

$$var[P_f(t_k)] = \frac{1}{N} \sum_j \sum_i (h_{ijk} w_{bh}(x_{ij}) - P_f(t_k))^2$$

$$cov[P_f(t_k)] = \frac{\sqrt{var[P_f(t_k)]/N}}{P_f(t_k)}$$



Adaptation Iteration 2/6





- n Calculate standard weights.
- n Effective sample size is 4, sufficient to update covariance matrix.
- New density is $N(\mu^*, \Sigma^*)$, generate samples and evaluate crack growth and response function.

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Adaptation Iteration 3/6





- > 100 samples used.
- ➤ All pofs and covs updated.

$$w_{bh}(x_{ij}) = \frac{q(x_{ij}, \theta_j)}{\sum_{l=1}^{N_m} (1 / N_m) q(x_{ij}, \theta_l)}$$

$$P_f(t_k) = \frac{1}{N} \sum_j \sum_i h_{ijk} w_{bh}(x_{ij})$$

$$var[P_f(t_k)] = \frac{1}{N} \sum_j \sum_i (h_{ijk} w_{bh}(x_{ij}) - P_f(t_k))^2$$

$$cov[P_f(t_k)] = \frac{\sqrt{var[P_f(t_k)]/N}}{P_f(t_k)}$$

- n Update balance heuristic importance weights.
- n Calculate estimates, estimator variances, and COVs.
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached.
- n Select time with the highest COV.



Adaptation Iteration 3/6





➢ New density added, N_{mix} = 6.
➢ 20 new samples generated.
➢ 120 samples total.



- n Calculate standard weights.
- n Effective sample size is 2, insufficient to update covariance matrix.
- New density is $N(\mu^*, I)$, generate samples and evaluate crack growth and response function.



Adaptation Iteration 4/6



120 samples used.All pofs and covs updated.

$$w_{bh}(x_{ij}) = \frac{q(x_{ij}, \theta_j)}{\sum_{l=1}^{N_m} (1 / N_m) q(x_{ij}, \theta_l)}$$

$$P_f(t_k) = \frac{1}{N} \sum_j \sum_i h_{ijk} w_{bh}(x_{ij})$$

$$var[P_f(t_k)] = \frac{1}{N} \sum_j \sum_i (h_{ijk} w_{bh}(x_{ij}) - P_f(t_k))^2$$

$$cov[P_f(t_k)] = \frac{\sqrt{var[P_f(t_k)]/N}}{P_f(t_k)}$$

- n Update balance heuristic importance weights
- n Calculate estimates, estimator variances, and COVs
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached
- n Select time with the highest COV



Adaptation Iteration 4/6







New density added, N_{mix} = 7.
20 new samples generated.
140 samples total.

$$\mu^{\star} = \frac{\sum_{j} \sum_{i} h_{ijk} w_{std}(\boldsymbol{x}_{ij}) \boldsymbol{x}_{ij}}{\sum_{j} \sum_{i} h_{ijk} w_{std}(\boldsymbol{x}_{ij})}$$
$$\Sigma^{\star} = \frac{\sum_{j} \sum_{i} (\boldsymbol{x}_{ij} - \boldsymbol{\mu}_{\star})^{\mathsf{T}} h_{ijk} w_{std}(\boldsymbol{x}_{ij}) (\boldsymbol{x}_{ij} - \boldsymbol{\mu}_{\star})}{\sum_{j} \sum_{i} h_{ijk} w_{std}(\boldsymbol{x}_{ij})}$$

- n Calculate standard weights
- n Effective sample size is 4, sufficient to update covariance matrix
- New density is $N(\mu^*, \Sigma^*)$, generate samples and evaluate crack growth and response function



Adaptation Iteration 5/6



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- Dydate balance heuristic importance weights.
- Calculate estimates, estimator variances, and COVs.
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached.
- n Select time with the highest COV.

- ➤ 140 samples used.
- All pofs and covs updated.
- > COV(t=0) now below threshold.





Adaptation Iteration 5/6



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- n Effective sample size is 6, sufficient to update covariance matrix.
- New density is $N(\mu^*, \Sigma^*)$, generate samples and evaluate crack growth and response function.







- n Update balance heuristic importance weights.
- Calculate estimates, estimator variances, and COVs.
- ⁿ Check exit condition (all COVs < ϵ_{cov}) or max iterations reached
- n All COVs < ϵ_{cov} .

- ➤ 160 samples used.
- > All COVs below threshold.
- CG results saved for all 160 samples.

$$w_{bh}(x_{ij}) = \frac{q(x_{ij}, \theta_j)}{\sum_{l=1}^{N_m} (1 / N_m) q(x_{ij}, \theta_l)}$$

$$P_f(t_k) = \frac{1}{N} \sum_j \sum_i h_{ijk} w_{bh}(x_{ij})$$

$$var[P_f(t_k)] = \frac{1}{N} \sum_j \sum_i (h_{ijk} w_{bh}(x_{ij}) - P_f(t_k))^2$$

$$cov[P_f(t_k)] = \frac{\sqrt{var[P_f(t_k)]/N}}{P_f(t_k)}$$



Academic Example Summary





- SFPOF computed using 160 samples.
- (60 initialization, 100 adaptation).
- All COVs below user-defined threshold of 20%.







	Typical Approach	AMIS Method
Probabilistic method	Numerical integration	Adaptive multiple importance sampling
Random variables	Only 3 random variables (toughness, initial crack size, max load)	Up to ~20 (toughness, initial crack size, max load, dadn variability, geometric factors)
Accuracy	Limit user control/information	Adaptivity ensures convergence to user-defined threshold
Reanalyses	Requires complete reanalysis	Mixture densities ensure efficient reanalysis
Efficiency	Efficient even for small POFs	Efficient even for small POFs



Transformation to Normal Space





- Transformation to standard normal space simplifies working with multiple distributions
 - Samples are generated in standard multivariate normal space
 - Nominal density is independent standard normal
 - Estimated optimal sampling density, likelihood ratio, and parameter updates use normal space
 - Samples are transformed to original space using inverse CDF transform (Nataf for correlated variables) for crack growth evaluation





Parameters $\hat{\mu}^{\star}(t)$, and $\hat{\Sigma}^{\star}(t)$ maximize the cross entropy

$$\boldsymbol{\theta}_{\text{opt}} = \operatorname{argmax}_{\boldsymbol{\theta}} \left(\sum_{i} \sum_{j} H(\boldsymbol{x}_{ij}, t) w(\boldsymbol{x}_{ij}) \ln \left(q(\boldsymbol{x}_{ij}, \boldsymbol{\theta}) \right) \right)$$

Closed form solution

n *H* depends on the Lincoln or Freudenthal formulation, e.g., $1 - F_{EVD}(\sigma_{RS}(\mathbf{x}, t_n))$



Initialization Overview



- Initialization algorithm searches for important regions for each time and adds component densities to the mixture
- n $N_{maxiter}$ and N_s set the maximum number of iterations and samples generated per iteration
- n $\{t_{pf}\}, CG(\cdot)$ and $H(\cdot)$ define the evaluation times, crack growth function and response function
- n c_{σ} controls the variance of the initialization sampling densities
- n ϵ_{KL} controls distance between initialization densities
 - Multiple steps will be taken for each evaluation time if the sampling density location changes with additional samples



Initialization Summary





- n Samples have been generated in and around the important regions
- Exploration by the initialization is very important the adaptation phase does not search for new important regions that are not found during the initialization phase

Mixture density weights – multiple definitions

 Large weights will not decrease as component densities are added

- Give the best weight out of all component densities for each sample
- If the mixture includes the nominal density, this ⁵⁸ weighting has an upper bound

Cross Entropy – Kullback-Leibler Divergence

Kullback-Leibler Divergence

$$\mathcal{D}(\underline{p(\mathbf{x})}, \underline{q(\mathbf{x})}) = \int \ln\left(\frac{p(\mathbf{x})}{q(\mathbf{x})}\right) p(\mathbf{x}) \, d\mathbf{x} = \int \ln(p(\mathbf{x})) \, p(\mathbf{x}) \, d\mathbf{x} - \int \ln(q(\mathbf{x})) \, p(\mathbf{x}) \, d\mathbf{x}$$

- n \mathcal{D} is a metric a measure of difference between two PDFs
 - $\mathcal{D} \ge 0$, $\mathcal{D} = 0$ when the PDFs are identical

Mixture density

$$\widehat{\text{POF}}(t) = \sum_{j=1}^{N_{mix}} \frac{1}{N_{samp}} \sum_{i=1}^{N_{samp}} H(\boldsymbol{x}_{ij}, t) \omega_j(\boldsymbol{x}_{ij}) \frac{f(\boldsymbol{x}_{ij})}{g(\boldsymbol{x}_{ij}, \theta_j)}$$

- n Mixture density is composed of component sampling densities optimal for individual times
- n Mixture importance weight is a weighted sum of importance weights from all component densities
- ⁿ The estimator is a double summation over component densities and samples from each density ⁶⁰

- When one or two weights have much larger weight, updating the covariance matrix parameters can give a point-like or degenerate sampling density
- n n_{eff} estimates the effective number of samples (between 0 and N) based on how evenly the weights are distributed 61
- ⁿ When n_{eff} is low, only the location parameter is updated

PDTA AMIS Algorithm

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Adaptation Overview

- n Adaptation algorithm adds sampling densities to the mixture until coefficient of variations (COVs) for all estimation times are below the user threshold, ϵ_{cov}
- Balance heuristic weights are used for to estimate POF and COV
- Standard weights are used for calculating new component density parameters
- n Inputs include outputs from the initialization stage

Add text why the 2 different weights are used Proof of consistency??

Single Evaluation Time Important Region Contour Comparison

- n Freudenthal important regions are much thinner
 - Reduced overlap which reduces potential for samples to contribute in more than one region

PDTA AMIS Reuse

- n Initial run (left plot) not including any inspections, completed with 880 samples
- The stored crack growth analyses were reevaluated with a modified response function including an inspection
- After re-running the adaptation algorithm the mixture density has been re-adapted using 320 additional samples to include the new important region near (0.01, 10)

Modeling Repair Quality using Weighted Branch Integration

- n Repair cracks account for the possibility that a new crack forms in the same component
- n The weighted branch integration method includes repairs as independent PDTA analyses
 - Each branch is weighted according to the percentage of cracks detected in the trunk and other branches

Repair Branch Analyses

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- ⁿ Branches are identical analyses except for the part from t = 17000 to t = 20000
- The PDTA AMIS algorithm is able to estimate POFs and PDETs for all branches from 1st branch samples

NASGRO Example with Inspections and Repairs

Parameter	Value
Width	Deterministic 2.5 in
Thickness	Deterministic 0.25 in
Initial Crack Size	<i>LN</i> (0.005, 0.002) in
Aspect Ratio (A/C) ¹	N(1.5, 0.14)
Fracture Toughness	N(34.8, 3.90) ksi √in
Log Paris Constant	N(-8.777, 0.08)
Paris Exponent	Deterministic 3.273
Hole Diameter	Deterministic 0.1562 in
Hole Offset ²	N(0.5, 0.05) in
Maximum Stress per Flight	<i>EVD</i> (16.74, 2.08, 0.0) ksi
Probability of Detection	<i>LN</i> (0.021, 0.028) in
Repair Ouality (crack size)	<i>LN</i> (0.01, 0.004) in

POF Results with Repairs

- n PDTA AMIS
 - Trunk inspected POF: 4060 samples
 - Trunk uninspected POF: +0 samples
 - Trunk Percent Cracks Det: +140 samples
 - Repair crack Branch POFs: 4060 samples

n 8260 total samples